

PATENT
Docket 13DV-13913

In the Specification

Please substitute the following paragraph for the corresponding paragraph being amended:

[0002] In a gas turbine engine, air is pressurized in a multistage axial compressor and mixed with fuel in a combustor for generating hot combustion gases which are discharged through several rows of turbine stages. A high pressure turbine powers the compressor, and a low pressure turbine powers a fan in a turbofan configuration for powering an aircraft in flight.